

# Elastodynamic analysis of torsion of shaft of revolution by line-loaded integral equation method\*

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**Abstract.** The dynamic response of an elastic torsion shaft of revolution is analysed by the Line-Loaded Integral Equation Method (LLIEM). A “Dynamic Point Ring Couple” (DPRC) is used as a fictitious fundamental load and is distributed in an elastic space along the axis of the shaft outside the shaft occupation. According to the boundary condition, our problem is reduced to a 1-D Fredholm integral equation of the first kind, which is simpler for solving than that of a 2-D singular integral equation of the same kind obtained by Boundary Element Method (BEM), for steady periodically varied loading. Numerical example of a shaft with quadratic generator under sinusoidal type of torque is given. Formulas for stresses and dangerous frequency are mentioned.

**Key words:** torsion of shaft of revolution; elastodynamic analysis; the line-loaded integral equation method.

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## 1. Introduction

The boundary element method (BEM) plays an important role in modern analysis, since it is better than the finite element method (FEM) for many problems. However, BEM is not the best for some problems. Many static problems have been analysed by LLIEM, for example, Yun (1979) solved the torsion problem of shaft of revolution; Yun (1981) analysed axially loaded pile embedded in an elastic half space, Yun *et al.* (1981) analysed ellipsoid compressed by two axial concentrated forces at two ends; Yun (1988) analysed the torsion problem of rigid circular shaft of varying diameter embedded in an elastic half space; Yun (1990) analysed the torsion problem of elastic shaft of revolution embedded in an elastic half space; Yun (1991) analysed rigid sloping pile under arbitrary loads; Yun and Su (1992) analysed a shaft (elastic cylindrical shell) with step-varied thickness, and/or with hinged supports, embedded in a granular half space under the action of gravity; Yun and Li (1995) analysed the in-plane-hinge-jointed rigid sloping piles. These examples show that LLIEM is efficient and simple in calculation. The major advantage of LLIEM over BEM is that LLIEM reduces a problem to a 1-D, non-singular integral equation (s), while BEM reduces the same problem to 2-D, singular integral equation (s) of the same kind. However, LLIEM has not been used for dynamic problem yet, this paper,

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presented at WCCM 3 (Yun 1994), extends the application of LLIEM to elastodynamic analysis of torsion shaft of revolution.

A general elastodynamic problem is a 4-D (three spatial dimensions plus time) problem. There are basically two approaches (direct and indirect methods) for elastodynamic analysis. In the direct method, the problem is solved directly in real time space while the indirect method the problem is firstly transformed and solved in an integral transformed space, and then the inverse transformation is taken to return the problem to its original space. Usually, the indirect method is recommender for the analysis of steady state dynamic problem due to its quasi-state manner. The indirect method usually has troubles in getting a solution of inverse integral transformation. The direct method usually spends more computation time since it repeats the calculation in each time domain, but it is easy to understand and are widely accepted by engineers.

There is no previous paper concerned with elastodynamic analysis of torsion shaft of revolution. The closest type of problem, in literature, related to our problem, is that dealing with elastodynamic analysis of timedomain BEM (Cheung, Lei & Tham 1993, Tham, Cheung & Lei 1994). However, these analyses are concerned with transient dynamic response of piles and half space, not for steady varied loading and not for shaft of revolution. In this paper, a steady varied loading problem of torsion shaft of revolution is analysed by LLIEM with both advantages of the direct method (e.g., easy to understand, without making integral transform and its inverse transform) and indirect method (less computation time, like a static problem).

In Sec. 2, as a fictitious fundamental load, the solution of DPRC at origin of an elastic space is derived, and its property is mentioned. In Sec. 3, the integral equation of our problem is derived, and the formulas of stress and dangerous frequency, which are interested to engineers, are obtained. Finally, numerical example of a shaft with quadratic generator under sinusoidal type torque is given.

## 2. The solution of "Dynamic Point Ring Couple" (DPRC) at origin

2.1. The solution of "Impulse Point Ring Couple" (IPRC) on  $z=0$  plane, at origin and  $t=0$ , in an elastic space

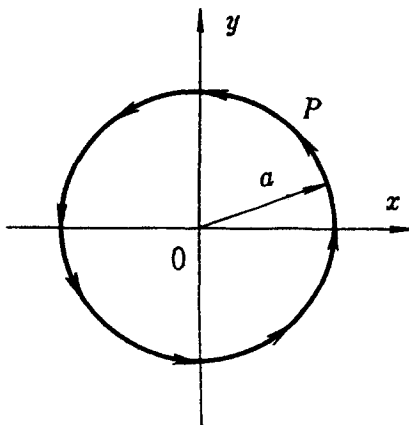


Fig. 1 An IPRC  $M_0\delta(0)$  at origin.  $M_0=\lim_{a\rightarrow 0} 2\pi a^2 P$

Fig. 1 shows an impulsive loading with uniform intensity  $P \cdot \delta(0)$  suddenly applied at time  $t=0$ , on a circle of  $z=0$  plane, with radius  $a$  and centered at origin of an elastic space, along the tangent of circumference. The limit of this loading system  $M_0\delta(0)$  for  $a \rightarrow 0$  is simply called an IPRC  $M_0\delta(0)$  at origin, where

$$M_0 = \lim 2\pi a^2 P \quad (a \rightarrow 0) \quad (1)$$

The  $i$ -th component of displacement field at point  $x$  and time  $t$  due to a unit concentrated impulse force acting at a point  $x_1$  and time  $t=0$  in the direction of the  $x_i$ -axis has been found (Eringen 1975) and is listed:

$$u_{ij}(x, t; x_1, 0) = \frac{1}{4\pi m} \left\{ \frac{t}{R^3} \left[ \frac{3x_i x_j}{R^2} - \delta_{ij} \right] \left[ H\left(t - \frac{R}{c_1}\right) - H\left(t - \frac{R}{c_2}\right) \right] \right. \\ \left. + \frac{x_i x_j}{R^3} \left[ \frac{\delta(t, R/c_1)}{c_1^2} - \frac{\delta(t, R/c_2)}{c_2^2} \right] + \frac{\delta_{ij} \delta(t, R/c_2)}{R c_2^2} \right\} \quad (2)$$

where  $R = |x - x_1|$ ,  $x_i = RR_{,i}$ ,  $x_j = RR_{,j}$ ,  $P$ -wave speed  $c_1$ ,  $S$ -wave speed  $c_2$  are:

$$c_1^2 = (\lambda + 2G)/m = 2G(1-\nu)/[m(1-2\nu)], \quad c_2^2 = G/m$$

$\lambda$ ,  $G$  are Lamé coefficients,  $m$ =density,  $\nu$ =Poisson's ratio,  $H$ =Heviside unit step function,  $\delta$ =Dirac function,  $\delta_{ij}$ =Kroncker delta.

The components of displacement and stress of a field point  $N(r, z)$  in cylindrical coordinates at time  $t$  due to a IPRC at origin and time  $t=0$  can be obtained using Eq. (2) by integration (see Yun and Gu 1993, for details) and are listed:

$$(U, V, W) = \left( 0, \frac{M_0 r}{8\pi G R_0^3} \left( \delta_{20} + \frac{R_0}{c_2} \dot{\delta}_{20} \right), 0 \right) \quad (3)$$

$$\sigma_r = \sigma_\theta = \sigma_z = \tau_{rz} = 0$$

$$\tau_{\theta r} = -\frac{3M_0 r^2}{8\pi R_0^5} \left[ \delta_{20} + \frac{R_0}{c_2} \dot{\delta}_{20} + \frac{1}{3} \frac{R_0^2}{c_2^2} \ddot{\delta}_{20} \right] \\ \tau_{\theta z} = -\frac{3M_0 r z}{8\pi R_0^5} \left[ \delta_{20} + \frac{R_0}{c_2} \dot{\delta}_{20} + \frac{1}{3} \frac{R_0^2}{c_2^2} \ddot{\delta}_{20} \right] \quad (4)$$

where and  $(\dot{\phantom{x}}) = \partial(\phantom{x})/\partial t$ ,  $\delta_{20} = \delta(t - R_0/c_2)$ ,  $R_0 = (r^2 + z^2)^{1/2}$ , and L'Hospital rule, Laplace transformation and inverse Laplace transformation have been used in the integration and limit process.

## 2.2. The solution of a DPRC at origin

According to the property of  $\delta$ -function, for arbitrary continuous function  $F(t)$ , we have

$$F(t) = \int_{-\infty}^{\infty} F(\tau) \delta(t - \tau) d\tau = \int_{-\infty}^{\infty} F(t - \tau) \delta(\tau) d\tau \quad (5)$$

by Eq. (5), we can obtain the components of displacement and stress of a field point  $N(r, z)$  at time  $t$  due to a DPRC at origin and time  $\tau$ , with time dependent magnitude  $M_0 F(\tau)$ . For example, the components of displacement of a field point  $N(r, z)$  at time  $t$  due to a DPRC at origin and time  $\tau$  with intensity  $M_0 \sin \omega \tau$  is: ( $0 \leq \omega \tau \leq 2\pi$ )

$$\begin{aligned} V &= \frac{M_0 r}{8\pi G R_0^3} \int_{-\infty}^{\infty} \sin \omega(t - \tau) [\delta_{20}(\tau) + R_0 c_2^{-1} \delta_{20}(\tau)] d\tau \\ &= \frac{M_0 r}{8\pi G R_0^3} \{ [\sin \omega(t - t_0) + \omega t_0 \cos \omega(t - t_0)] H_{20} + t_0 \sin \omega(t - t_0) \cdot \Delta \delta_{20} \} \end{aligned} \quad (6)$$

where  $H_{20} = H(t - t_0)$ ,  $t_0 = R_0 / c_2$ ,  $\Delta \delta_{20} = \lim_{\varepsilon \rightarrow 0} \{ \delta(t + \varepsilon - t_0) - \delta(t - \varepsilon - t_0) \}$ ,  $\omega$  = frequency. In the derivation of Eq. (6),

$$\int_{-\infty}^{\infty} = \int_{-\infty}^{t_0 - \varepsilon} + \int_{t_0 - \varepsilon}^{t_0 + \varepsilon} + \int_{t_0 + \varepsilon}^{\infty}$$

the formular of integral by parts and mid-value theorem of integration have been used. The last term involved  $\Delta \delta_{20}$  belongs to the type of  $(\infty - \infty)$ , whether it would be infinite or finite, it just represents the response at  $t = t_0$ , and disappears for  $t \neq t_0$ . For a torsion problem of periodical loading, where the instant response is not so important comparing with the long lasting term  $H_{20}$ . Therefore, we neglect the instant terms in the following for simplifying, and list the components of displacement and stress

$$V = [M_0 r / (8\pi G R_0^3)] [\sin \omega(t - t_0) + t_0 \omega \cos \omega(t - t_0)] H_{20} \quad (7)$$

$$\tau_{\theta r} = -3M_0 r^2 \phi(y_0, y) / [8\pi R_0^5], \quad \tau_{\theta z} = -3M_0 r z \phi(y_0, y) / [8\pi R_0^5] \quad (8)$$

$$\phi(y_0, y) = A(y_0) \sin(y - y_1) \cdot H_{20} \quad (9)$$

$$A(y_0) = [(1 - y_0^2/3)^2 + y_0^2]^{1/2}, \quad y = \omega t, \quad y_0 = \omega t_0, \quad t_0 = R_0 / c_2 \quad (10)$$

$$y_1 = t g^{-1} \{ [(1 - y_0^2/3) \sin y_0 - y_0 \cos y_0] / [(1 - y_0^2/3) \cos y_0 + y_0 \sin y_0] \} \quad (11)$$

Two properties of the solution Eqs. (7) and (8) are mentioned:

1. For any instant, we have

$$\tau_{\theta r} / \tau_{\theta z} = r / z, \quad \text{for } \forall t \quad (12)$$

Eq. (12) shows that a family of zero stressed cone surface with the same apex at origin and the  $z$ -axis as its symmetrical axis appears in the elastic space at any time.

2. The solution Eqs. (7) and (8) satisfies the differential equation of motion:

$$\partial \sigma_r / \partial r + (1/r) \partial \tau_{\theta r} / \partial \theta + \partial \tau_{rz} / \partial z + (\sigma_r - \sigma_{\theta}) / r = m \ddot{U} - F_r \quad (13a)$$

$$\partial \tau_{\theta r} / \partial r + (1/r) \partial \sigma_{\theta} / \partial \theta + \partial \tau_{\theta z} / \partial z + 2\tau_{\theta r} / r = m \ddot{V} - F_{\theta} \quad (13b)$$

$$\partial \tau_{rz} / \partial r + (1/r) \partial \tau_{\theta z} / \partial \theta + \partial \sigma_z / \partial z + \tau_{rz} / r = m \ddot{W} - F_z \quad (13c)$$

Obviously, Eqs. (7) and (8) satisfy Eqs. (13a) and (13c) if the body force  $F$  is neglected. By Eq. (9), we have

Substituting the above relationship, Eqs. (7) and (8) into Eq. (13b), we get  $\partial\tau_{\theta r}/\partial r + \partial\tau_{\theta z}/\partial z + 2\tau_{\theta r}/r = -3M_0 r/(8\pi R_0^3) = m\ddot{V}$ .

### 3. Formulation of the integral equation

Suppose that the shaft of revolution is located in  $z \in [0, b]$ . Let the fictitious load DPRC with unknown intensity  $x(s) \sin \omega t$  to be distributed along the  $s(= -z)$ -axis in  $[a, L](b \ll L < \infty)$ . The reason for DPRC distribution outside the shaft occupation is to agree with the fact that no singularity occurs at points of the axis of symmetry of a torsion shaft. Usually, an inhomogeneous integral equation is easier to obtain a numerical solution than that of its homogeneous case. In order to reduce our problem to an inhomogeneous integral equation, analogous to an adding term of uniformly flow in the method of source-sink distribution in fluid mechanics, besides the above DPRC distribution, a single DPRC with intensity  $C \cdot s_0^4 \sin \omega t$  ( $C$ -constant) acting at  $s=s_0$  ( $s_0 \gg b$ ) is added (Fig. 2). The stress at a point  $N(r, z)$  in the elastic space and at time  $t$ , due to the above all DPRC, by Eq. (8), is

$$\tau_{\theta_r} = -[3r/(8\pi)] \left[ \int_{\theta}^L (z+s) \phi(y_s, y) R_s^{-5} x(s) ds - C \cdot \phi(y_{s_0}, y) \right] \quad (14b)$$

where  $R_s=[r^2+(z+s)^2]^{1/2}$ ,  $R_{s_0}=[r^2+(z+s_0)^2]^{1/2}$ ,  $y_s=\omega t_s$ ,  $y_{s_0}=\omega t_{s_0}$ ,  $t_s=R_s/c_2$ ,  $t_{s_0}=R_{s_0}/c_2$ ,  $y$ ,  $y_s$ ,  $y_{s_0} \in [0, 2\pi]$ ,  $\phi(y_s, y)$  and  $\phi(y_{s_0}, y)$  are the same expression as  $\phi(y_0, y)$  shown in Eq. (9) but instead of  $y_0$  by  $y_s$  and  $y_{s_0}$  respectively. And  $s_0$  ( $L > s_0 \gg b$ ) is chosen in the following.

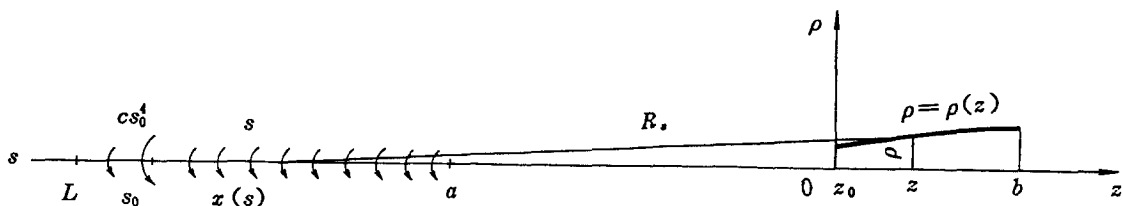


Fig. 2 DPRC distributed in  $[a, L]$  which produces a zero stressed surface of revolution  $\rho=\rho(z)$  in the elastic space at any time

By stress analysis, the normal direction  $n$  of the zero stressed plane in a small cylindrical element of a field point  $N(r, z)$  is determined by

$$\operatorname{tg} \alpha = \tau_{\theta r} / \tau_{\theta z}$$

where  $\alpha + \pi/2$  is the angle between  $n$  and the  $z$ -axis. Every element of points has such a zero stressed plane, therefore a family of zero stressed surfaces of revolution occurs in the elastic space. One of them passing through  $N(r, z)$  is constructed to coincide with the given generator  $\rho = \rho(z)$ , where the derivative  $\rho' = d\rho/dz$  is equal to

$$\rho' = d\rho/dz = \tau_{\theta r} / \tau_{\theta z} \quad (15)$$

Substituting Eq. (14) into Eq. (15), we have

$$C \rho' \phi(y_{s0}, y) = \int_a^L [\rho - \rho'(z+s)] \phi(y_s, y) R_s^{-5} x(s) ds \quad (16)$$

By the theorem of integral mean value of function, there exists a  $\phi(y_{s0}, y)$  ( $\phi(y_{sa}, y) \leq \phi(y_{s0}, y) \leq \phi(y_{sl}, y)$ ) such that

$$\phi(y_{s0}, y) \int_a^L [\rho - \rho'(z+s)] R_s^{-5} x(s) ds = \int_a^L [\rho - \rho'(z+s)] R_s^{-5} \phi(y_s, y) x(s) ds \quad (17)$$

where  $y_{sa} = \omega R_{sa} / c_2$ ,  $R_{sa} = [\rho^2 + (z+a)^2]^{1/2}$ , and so on.

Now,  $s_0$  is chosen by Eq. (17), and substituting Eq. (17) into Eq. (16), we have

$$C \rho' = \int_a^L [\rho - \rho'(z+s)] R_s^{-5} x(s) ds \quad (18)$$

Since constant  $C$  and generator  $\rho = \rho(z)$  are given (without loss of generality, let  $C=1$ ), then, Eq. (18) is a Fredholm integral equation of the first kind. Obviously, Eq. (18) is independent of time. This means that the zero stressed surface of revolution with generator  $\rho = \rho(z)$  appears in the elastic space and keeps unchange at any time. This characteristic coincides with the zero stressed boundary condition of dynamic torsion of shaft of revolution. Thus the above solution can be regarded as the solution of dynamic torsion of shaft of revolution. Eq. (18) also shows that the solution  $x(s)$ , like a static case, is wholly determined by the given generator.

Once  $x(s)$  of Eq. (18) has been solved, the stresses can be calculated by Eq. (14). The relation between  $x(s)$  and the given dynamic torque  $M_T(z_0) = T_0 \sin \omega(t - t_i)$  can be obtained by

$$T_0 \sin \omega(t - t_i) = M_T(z_0) = 2\pi \int_0^{\rho(z_0)} \tau_{\theta z} r^2 dr \quad (19)$$

Substituting Eq. (14b) with mid-value  $\phi(y_{s0}, y)$  into Eq. (19), and again using the theorem of integral mean value of function, there exists a  $r_m$  ( $0 < r_m < \rho$ ) in  $z = z_0$  such that

$$\phi(y_{m0}, y) K(z_0) = \int_0^{\rho(z_0)} \left[ \int_a^L (z_0 + s) R_s^{-5} x(s) ds - c \right] r^3 \phi(y_{s0}, y) dr \quad (20)$$

then, we have

$$T_0 \sin \omega(t - t_i) = -(3/4) A (y_{m0}) K(z_0) \sin(y - y_{1m}) \quad (21)$$

where  $y_{m0} = \omega t_{m0}$ ,  $t_{m0} = R_{m0} / c_2$ ,  $R_{m0} = [r_m^2 + (z_0 + s_0)^2]^{1/2}$ ,

$$K(z_0) = \int_0^{\rho(z_0)} \left[ \int_a^L (z_0 + s) R_s^{-5} x(s) ds - c \right] r^3 dr \quad (22)$$

$A(y_{m0})$  and  $y_{1m}$  are the same expressions as Eqs. (10) and (11), instead of  $y_0$  by  $y_{m0}$ , respectively. From Eq. (21), we have

$$T_0 = -(3/4)A(y_{m0})K(z_0) \quad (23)$$

$$t_i = y_{1m}/\omega \quad (24)$$

The maximum stress is of interest to engineers. Suppose that  $2\rho(z_0)$  is the maximum diameter of the shaft, then, the maximum stresses of the shaft, by Eq. (14), are:

$$\tau_{\theta z}(\rho, z_0) = -(3\rho/(8\pi))f_1(\rho)\phi(y_{s0}, y) \quad (25a)$$

$$\tau_{\theta r}(\rho, z_0) = -(3\rho^2/(8\pi))f_2(\rho)\phi(y_{s0}, y) \quad (25b)$$

$$f_1(\rho) = \int_a^L (z_0 + s) R_{s\rho}^{-5} x(s) ds - C, \quad R_{s\rho} = [\rho^2 + (z_0 + s)^2]^{1/2}, \quad f_2(\rho) = \int_a^L R_{s\rho}^{-5} x(s) ds.$$

Although  $x(s)$  is independent of dynamic factor, but stress varies with frequency  $\omega$  and time  $t$ . The so-called “dangerous” frequency  $\omega_d$  and time  $t_d$ , at which  $\tau_{\theta z}(\rho, z_0)$  and  $\tau_{\theta r}(\rho, z_0)$  attend to maximum, can be found by solving the following optimization problem (O.P.):

O.P.: Find  $\omega$  and  $t$  such that

$$\begin{aligned} &\text{Max } \tau_{\theta z}(\rho, z_0) \text{ and } \tau_{\theta r}(\rho, z_0) \text{ of Eq. (25)} \\ &\text{s.t. Eq. (23), i.e., } A(y_{m0}) = -(4/3)T_0/K(z_0) = C_1 \end{aligned}$$

This is a constrained O.P., it can be solved as an un-constrained O.P. by the method of Lagrange multiplier. Let

$$\Psi = \phi(y_{s0}, y) + \beta[A(y_{m0}) - C_1] \quad (26)$$

where  $\beta$  is the Lagrange multiplier. Then

$$\partial\Psi/\partial\omega = (\partial\Psi/\partial y_{s0})(\partial y_{s0}/\partial\omega) = 0 \quad (27)$$

$$\partial\Psi/\partial t = (\partial\Psi/\partial y)(\partial y/\partial t) = 0 \quad (28)$$

$$\partial\Psi/\partial\beta = 0 \quad (29)$$

$$\text{we get: } y - y_1 = \pi/2 \quad (30)$$

$$\beta = -t_{s0}/t_{m0} \quad (31)$$

$$A(y_{m0}) = C_1 \quad (32)$$

From Eq. (32),

$$T_0 = -(3/4)K(z_0)A(y_{m0}), \quad (0 \leq y_{m0} \leq 2\pi) \quad (33)$$

Eq. (33) shows that the amplitude of dynamic torque  $T_0$  related to  $y_{m0} = \omega_d t_{m0}$ , by Eq. (10), where

$$\omega_d = y_{m0}/t_{m0} = 2\pi/t_{m0} \quad (34)$$

$$T_0 = T_{0max} = -(3/4)K(z_0)A(2\pi) \quad (35)$$

From Eq. (30), we get  $t_d = t_1 + t_{m0}/4$ , however,  $t_d$  is not important, while  $\omega_d$  is interesting to engineers. Eq. (34) shows that the dangerous frequency related to  $t_{m0} = R_{m0}/c_2$ , which relates to the given generator  $\rho = \rho(z)$ .

#### 4. Numerical solution and example

Eq. (18) is rewritten simply as

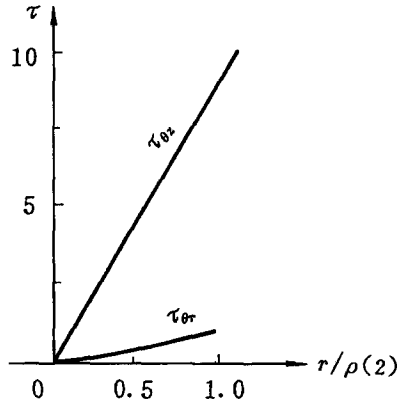


Fig. 3 The distribution of stresses on cross section  $z=2$  at  $t=300$  sec

$$\int_a^L K(z, s) x(s) ds = F(z) \quad (36)$$

$$K(z, s) = [\rho - \rho'(z + s)] R_s^{-5}, \quad F(z) = C \rho'$$

Eq. (36) is solved by its discrete form, i.e.,

$$\begin{aligned} \sum_{j=1}^n K_{ij} x_j &= F_i, \quad i, j = 1, 2, \dots, n. \\ x_j &= x(s_j), \quad F_i = F(z_i), \quad K_{ij} = \int_{(j-1)\Delta L}^{j\Delta L} K(z_i, s) ds, \\ \Delta L &= (L - a)/n, \quad s_j = j \cdot \Delta L, \quad z_i = i \cdot (b - z_0)/n. \end{aligned} \quad (37)$$

Once  $x_j$  has been solved from Eq. (37), the stress and the corresponding torque  $M_T(z_0)$  can be numerically calculated by Eq. (14) and Eq. (19), where the integral is replaced by summation, respectively.

Numerical example of a dynamic torsion shaft of revolution with generator  $\rho = \rho(z) = \sqrt{1 + 0.1z}$  located in  $z_0=0$ ,  $b=5$  as shown in Fig. 2. In this example,  $m=7.8 \times 10^3$  (Pa.sec<sup>2</sup>/M<sup>2</sup>),  $G=8.0 \times 10^{10}$  (Pa),  $a=10$  (M),  $L=20$  (M),  $n=20$  are used.

The distributions of stresses on different cross sections, at different time, due to dynamic torque  $M_T(0)$  at  $z_0=0$  are plotted and shown to be similar in pattern. A typical pattern of distribution of stresses on cross section  $z=2$  (M) is presented in Fig. 3. In which the stress  $\tau_{\theta z}$  is almost linear and  $\tau_{\theta r}$  is obviously far less than  $\tau_{\theta z}$ . This result approaches to that of a torsion

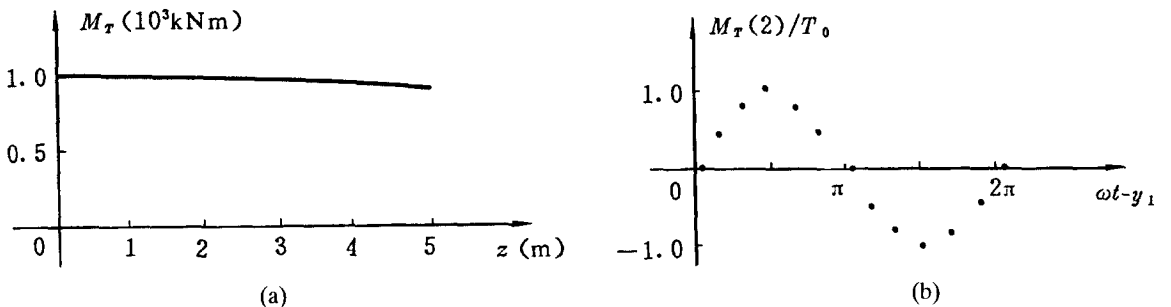


Fig. 4 (a) Torque of different cross section at  $t=300$  sec. (b) Torque of cross section  $z=2$  at different time



cylinder and is expected, since the shaft  $\rho = \sqrt{1 + 0.1z}$  is slowly varied in diameter. Fig. 4 shows the torque of different cross sections at the same time and the torque of the same cross section at different time.

The ratio of maximum amplitude of dynamic torque  $T_{0max}$  due to dangerous frequency  $\omega_d$  of Eq. (34), to its minimum amplitude  $T_{0s}$  due to  $\omega=0$  (the static case), by Eqs. (35) and (23), is

$$T_{0max}/T_{0s} = (1 + y_{m0}^2/3 + 4_{m0}^4/9)^{1/2} \approx y_{m0}^2/3 = (2\pi)^2/3 = 13.0 \quad (38)$$

This result shows that the effect of dynamic action is of significance.

## 5. Conclusions

The problem of dynamic torsion of shaft of revolution, twisted by end torque  $T_0 \sin \omega(t - t_i)$ , is studied by LLIEM with both advantages of the direct method (e.g., easy to understand, without making integral transform and its inverse transform) and indirect method (less computation time, like a static problem). In which, a zero stressed surface of revolution with given generator  $\rho = \rho(z)$  can be constructed to appear in an elastic space at any time by DPRC, with unknown intensity  $(x(s) + C \cdot s_0^4) \sin \omega t$ , distributing on  $[a, L]$  ( $L > a \gg b$ ) along the  $s(= -z)$ -axis outside the shaft occupation. Then, the problem is reduced to a Fredholm integral equation of the first kind and can be solved by known methods.

The advantage of LLIEM over BEM for dynamic analysis is the obtained integral equation is 1-D and non-singular, while the BEM needs to treat singularity (e.g.,  $0(1/r)$  and  $0(1/r^2)$ ), see, Y. K. Cheung *et al.* 1993) by special method.

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